ADVANCED ELECTRIC PROPULSION FOR FAST MANNED MISSIONS TO MARS AND BEYOND

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One of the most serious challenges of human exploration of Mars lies in the long transfer time required, at least 180 days using chemical or nuclear thermal propulsion. A significant reduction in transfer time would relieve many issues, both technical and psychological, for human Mars missions.

High-power electric propulsion systems are probably the best candidate for near-future fast manned Mars missions. They can provide a specific impulse as high as 10 times the one given by chemical propulsion, leading to a significant reduction of the propellant mass. However, this translates into a significant reduction of the transfer time only if the mass of the power source is kept well below 10 kg/kW.

High-power nuclear electric power systems are considered as possible solutions to this problem. Today a space nuclear reactor can be realized with a specific mass of 30 kg/kW; however in the medium term, using advanced power-generation technologies that, while challenging, are considered feasible, for example a high-temperature Rankine conversion cycle, it could be possible to reach a specific mass of 5 kg/kW. This would allow a transfer time of about 120 days, 60 days less than the best transfer time with chemical propulsion.

A very fast Earth-to-Mars transfer in less than 120 days appears to be unrealistic in the medium term because it depends on a hypothetical breakthrough with the nuclear electric power source, requiring a specific mass that would typically be less than 1 kg/kW.

This paper also proposes an advanced electric propulsion concept, Laser-powered Electric Propulsion (LEP), where the nuclear reactor is replaced by a light-weight photovoltaic (PV) collector. A high-power laser beam from an inspace laser power source is aimed at the PV collector on the target spacecraft, where it is converted to electric power for an advanced electric propulsion system.

The PV collector/converter on the spacecraft can be tuned to the laser wavelength, thus achieving high monochromatic conversion efficiencies, currently $\sim 50\%$ with the potential to reach 80% in the near future. Such a light-weight power source could have a specific mass of less than 1 kg/kW, enabling very fast manned missions to Mars and beyond.

References:

Goebel, D. M. and Katz, I., "Fundamentals of Electric Propulsion: Ion and Hall Thrusters", JPL/Wiley press, 2008 Berend, N., et al., "How fast can we go to Mars using high power electric propulsion?", 48th AIAA Joint Propulsion Conference, AIAA-2012-3889, Atlanta, Georgia, 2012

Forward, R. L., "Advanced Propulsion Concepts Study: Comparison of SEP and Laser Electric Propulsion", Final Report, JPL Contract 954085, June 1985.

Bett, A. W., Dimroth, F., Lockenhoff, R., Oliva, E., and Schubert, J., "III-V solar cells under monochromatic illumination," 33rd IEEE Photovoltaic Specialists Conference, 2008

Genovese, A., "Advanced Electric Propulsion for Interstellar Precursor Missions", book chapter in "Beyond the Boundary", edited by K. Long and published by I4IS, 2014

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